OPTIMAL CONTROL SYSTEMS Examples

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OUTLINE

Recap

 $STATE \ CONSTRAINTS$

OBSTACLE AVOIDANCE

MOON LANDER



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PONTRYAGIN MINIMAL PRINCIPLE

Necessary conditions for an optimal control:

• for all $t \in [t_1, t_2]$,

$$\begin{split} \dot{x}^* &= H_{\lambda}^T \left(x^*, \lambda^*, u^* \right) \\ \dot{\lambda}^T &= -H_x \left(x^*, \lambda^*, u^* \right) \\ H \left(x^*, \lambda^*, u^* \right) &\leq H \left(x^*, \lambda^*, u \right), \quad \forall u \in U \\ H \left(x^*, \lambda^*, u^* \right) &= c \end{split}$$

- with boundary conditions:
 - initial state $x(t_1) = x^1$
 - transversality condition $(\ell_x \lambda^T)_{t_2} \delta x(t_2) = 0$
 - if the terminal time is free, then $H(x^*, \lambda^*, u^*)_{t_2} = 0$



PRELIMINARY EXAMPLE

Let us reconsider the earlier problem, but with control constraint:

 $\dot{x}_1 = x_2$ $\dot{x}_2 = -x_2 + u, \quad |u| \le 1$

We wish to steer from an arbitrary initial state to the origin and minimize the cost

$$J = \int_0^{t_2} \left(1 + \frac{1}{2}u^2\right) dt$$

As before

$$H = 1 + \frac{1}{2}u^{2} + \lambda_{1}x_{2} + \lambda_{2}(-x_{2} + u)$$

Thus,

$$\dot{\lambda}_1=0, \quad \dot{\lambda}_2=-\lambda_1+\lambda_2$$

But the control is

$$u^* = \arg \min_{|u| \le 1} \left\{ \frac{1}{2}u^2 + \lambda_2 u \right\}$$



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PRELIMINARY EXAMPLE, CONT'D

Thus we compute the control:

$$u^* = \begin{cases} -1 & \lambda_2 > 1\\ -\lambda_2 & -1 \le \lambda_2 \le 1\\ 1 & \lambda_2 < -1 \end{cases} = -\operatorname{sat}\lambda_2$$

We also have the terminal condition

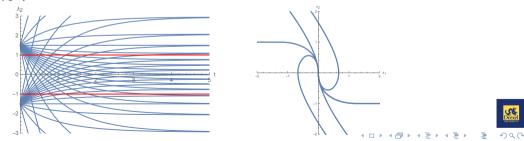
$$H_{t_2} = 1 + \frac{1}{2} \operatorname{sat}^2 \lambda_2(t_2) - \lambda_2(t_2) \operatorname{sat} \lambda_2(t_2) = 0 \Rightarrow \lambda_2(t_2) = \pm 3/2$$

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PRELIMINARY EXAMPLE, CONT'D

$$H_{t_2} = 1 + \frac{1}{2} \operatorname{sat}^2 \lambda_2 (t_2) - \lambda_2 (t_2) \operatorname{sat} \lambda_2 (t_2) = 0 \Rightarrow \lambda_2 (t_2) = \pm 3/2$$
$$\left(\ell_x - \lambda^T \right)_{t_2} \delta x (t_2) = 0 \Rightarrow \lambda^T (t_2) \left. \frac{\partial \phi(x)}{\partial x} \right|_{x=x^*}, \phi(x) = x_1^2 + x_2^2 - \varepsilon^2 = 0$$

PlotList = Join[Range[$\pi/20$, $\pi(1-1/20)$, $\pi/20$], Range[$-\pi/20$, $-\pi(1-1/20)$, $-\pi/20$]]; gr1 = Plot[Map[(($\{\lambda 2[t]\}$ /. sol λ) /. $\sigma \rightarrow \#$) &, PlotList], {t, 0, 5}, PlotRange \rightarrow {{0, 5}, {-3, 3}}, AxesLabel \rightarrow {"t", " λ_2 "}]; $gr2 = Plot[{1, -1}, {t, 0, 5}, PlotStyle \rightarrow {{Red}}];$ Show[gr1, gr2]



STATE CONSTRAINTS

- Here, we consider problems in which the state is restricted to a subset of $S \subset R^n$.
- It will be assumed that the the allowable domain can be defined by a set of inequalities

$$\phi_1(x,t) \ge 0, \ldots, \phi_s(x,t) \ge 0$$

where each ϕ_i is a smooth function of *x*, *t*.

lntroduce a new state variable $x_{n+1}(t)$, defined by

 $\dot{x}_{n+1}(t) = \phi_1^2(x,t) u_0(-\phi_1) + \dots + \phi_s^2(x,t) u_0(-\phi_s), \quad u_0 \text{ denotes the unit step}$

with boundary conditions

$$x_{n+1}(t_1) = 0, x_{n+1}(t_2) = 0$$

note that these boundary conditions can be satisfied only if the constraints are satisfied along the entire trajectory.

• The necessary condition stated above can be applied, with the additional state equation.



SATISFYING THE NECESSARY CONDITIONS

Given an initial state there are two possibilities:

- There does not exist any trajectory that satisfies all of the necessary conditions including the state constraints
- There exists one or more optimal trajectories, and these can be of two types
 - the entire trajectory lies interior to the state constraint set *S*
 - trajectory segments of finite length lie on the boundary of S



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MOTION ON THE CONSTRAINT BOUNDARY

Now, we will consider optimal trajectory segments on the constraint boundary. Suppose

the allowable region $S \subset R^n$ is characterized by the scalar-valued function $\phi(x)$:

 $S = \{x \in \mathbb{R}^n \, | \phi(x) \le 0\}$

Define

$$c(x,u) = \frac{\partial \phi(x)}{\partial x} f(x,u)$$

If $x(t) \in \partial S$ for some finite time interval $t_a \leq t \leq t_b$ then $c(x, u) \equiv 0$ on $t_a \leq t \leq t_b$. This simply means that \dot{x} lies in the tangent plane to the surface ∂S .



OPTIMAL MOTION ON THE CONSTRAINT BOUNDARY

LEMMA (MINIMAL PRINCIPLE ON BOUNDARY)

Suppose $u^*(t)$ and $x^*(t)$ are optimal and $x^*(t) \in \partial S$ for $t_a \leq t \leq t_b$. Then there exists adjoint variables $\lambda(t)$ and a scalar valued function $\alpha(t)$ such that on $t \in [t_a, t_b]$

$$\begin{split} \dot{\lambda}^* &= -\frac{\partial H\left(x^*, \lambda^*, u^*\right)}{\partial x} - \alpha \frac{\partial c\left(x^*, u^*\right)}{\partial x} \\ H\left(x^*, \lambda^*, u^*\right) &= \min_{u \in U} \left\{ H\left(x^*, \lambda^*, u\right) | c\left(x, u\right) = 0 \right\} \\ u^*, \alpha^* &= \arg\min_{u \in U, \alpha \in \mathbb{R}} \left\{ H\left(x^*, \lambda^*, u\right) + \alpha c\left(x^*, u\right) \right\} \end{split}$$



REMARKS

- If x* (t) ∈ intS for times t₁ ≤ t < t_a, then the ordinary minimum principle holds therein.
- The adjoint variables are continuous at t_a , i.e.,

$$\lambda\left(t_{a}^{-}\right) = \lambda\left(t_{a}^{+}\right)$$

However, when leaving the boundary

$$\lambda\left(t_{b}^{+}\right) = \lambda\left(t_{b}^{-}\right) - \alpha\left(t_{b}\right)\frac{\partial\phi\left(x\left(t_{b}\right)\right)}{\partial x}$$



(a)

EXAMPLE

Consider the problem of motion in a plane defined dynamics

 $\dot{x}_1 = u_1$ $\dot{x}_2 = u_2$

with control constraint

$$U = \left\{ u \in R^2 \left| u_1^2 + u_2^2 - 1 \le 0 \right. \right\}$$

We seek to steer an arbitrary state to the origin in minimum time

$$J = \int_{t_1}^{t_2} dt$$

while avoiding the obstacle - a unit circle centered at (2,0). The admissible space is defined by

$$\phi(x,t) = 1 - (x_1 - 2)^2 - x_2^2 \le 0$$



(a)

EXAMPLE, CONT'D

The ordinary minimum principle which holds while the trajectory is in the interior of *S*:

$$H(x,\lambda,u) = 1 + \lambda_1 u_1 + \lambda_2 u_2$$

so that

$$u_1^* = -\frac{\lambda_1}{\sqrt{\lambda_1^2 + \lambda_2^2}}, \quad u_2^* = -\frac{\lambda_2}{\sqrt{\lambda_1^2 + \lambda_2^2}}$$

and the adjoint equations are

$$\dot{\lambda}_1 = 0, \quad \dot{\lambda}_2 = 0$$

and we require the free time condition

$$H(t_2) = \lambda_1^2(t_2) + \lambda_2^2(t_2) - 1 = 0 \Rightarrow \lambda_1(t_2), \lambda_2(t_2) \text{ are on a unit circle in } \lambda - \text{space}$$



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EXAMPLE CONT'D

Without the obstacle we have

$$\begin{aligned} \lambda_1 \left(t \right) &= \cos \theta, \quad \lambda_2 \left(t \right) = \sin \theta \\ u_1^* &= -\cos \theta, \quad u_2^* &= -\sin \theta \end{aligned}$$

All trajectories are straight lines!

Now suppose that $x^*(t) \in \partial S$ for $t_a \leq t \leq t_b$. Compute

$$c(x, u) = -2(x_1 - 2)u_1 - 2x_2u_2 = 0$$

and

$$\dot{\lambda}_1 = 2\alpha u_1, \quad \dot{\lambda}_2 = 2\alpha u_2$$



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EXAMPLE, CONT'D

Now, we need to find u^* by minimizing *H* subject to the constraint c(x, u) = 0.

 $\min_{u\in U,\alpha\in R}\left\{H\left(x,\lambda,u\right)+\alpha c\left(x,u\right)\right\}$

First, consider a transformation

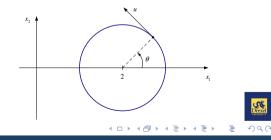
$$x_1 \rightarrow 2 + \cos \theta, \ x_2 \rightarrow \sin \theta$$

so that points on ∂S are parameterized by θ as shown in the figure.

There are two solutions

$$u_1 = -\sin\theta, \ u_2 = \cos\theta, \alpha = (\lambda_1 \cos\theta + \lambda_2 \sin\theta) / 2 u_1 = \sin\theta, \ u_2 = -\cos\theta, \alpha = (\lambda_1 \cos\theta + \lambda_2 \sin\theta) / 2$$

The first minimizes *H* on the upper half circle and the second on the lower half circle.



EXAMPLE: OBSTACLE APPROACH

Let us focus on the top half circle. Consider the two relations

$$H^* = 1 + \lambda_1 u_1 + \lambda_2 u_2 \equiv 0, \quad \alpha = \left(\lambda_1 u_2 - \lambda_2 u_1\right)/2$$

Use these equations and the first of the adjoint differential equations

$$\dot{\lambda}_1 = 2\alpha u_1$$

to obtain

$$\dot{\lambda}_1 = -\lambda_1 \tan \theta + \sin \theta \tan \theta$$

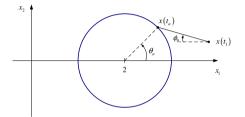
 $\lambda_2 = \lambda_1 \tan \theta - \sec \theta$

Now, solve the differential equation first to obtain

$$\lambda_1 = c_1 \cos \theta - \theta \cos \theta + \sin \theta$$
$$\lambda_2 = c_1 \sin \theta - \theta \sin \theta - \cos \theta$$



EXAMPLE: OBSTACLE APPROACH, CONT'D



Continuity of the adjoint variables yields

$$\lambda_1 (t_a^-) = \lambda_1 (t_a^+) \Rightarrow \cos \phi_0 = c_1 \cos \theta_a - \theta_a \cos \theta_a + \sin \theta_a$$
$$\lambda_2 (t_a^-) = \lambda_2 (t_a^+) \Rightarrow -\sin \phi_0 = c_1 \sin \theta_a - \theta_a \sin \theta_a - \cos \theta_a$$
$$c_1 = \theta_a, \quad \theta_a + \phi_0 = \frac{\pi}{2}$$



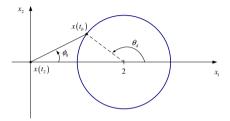
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EXAMPLE: OBSTACLE DEPARTURE



Just before departure we have

$$\lambda_1 \left(t_b^- \right) = \theta_a \cos \theta_d - \theta_d \cos \theta_d + \sin \theta_d \lambda_2 \left(t_b^- \right) = \theta_a \sin \theta_d - \theta_d \sin \theta_d - \cos \theta_d$$

Just after departure, since the usual equations obtain, we have

$$\lambda_1 \left(t_b^+
ight) = \cos \phi_0, \quad \lambda_2 \left(t_b^+
ight) = \sin \phi_0$$



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EXAMPLE: OBSTACLE DEPARTURE, CONT'D

Now, we need to satisfy the departure continuity equations

$$\lambda\left(t_{b}^{+}\right) = \lambda\left(t_{b}^{-}\right) - \alpha\left(t_{b}\right)\frac{\partial\varphi\left(x\left(t_{b}\right)\right)}{\partial x}$$

Compute:

$$\frac{\partial \varphi}{\partial x}\Big|_{\theta_d} = \begin{bmatrix} -2(x_1 - 2) & -2x_2 \end{bmatrix} = \begin{bmatrix} -2\cos\theta_d & -2\sin\theta_d \end{bmatrix}$$
$$\alpha|_{\theta_d} = \frac{1}{2}\left(\lambda_1\left(t_d^{-}\right)\cos\theta_d + \lambda_2\left(t_d^{-}\right)\sin\theta_d\right) = \frac{\theta_a - \theta_d}{2}$$

Consequently,

$$\lambda_1 \left(t_b^+ \right) = \lambda_1 \left(t_b^- \right) - \alpha \left(t_b \right) \frac{\partial \varphi(x(t_b))}{\partial x_1} \Rightarrow \cos \phi_0 = \sin \theta_d$$

$$\lambda_2 \left(t_b^+ \right) = \lambda_2 \left(t_b^- \right) - \alpha \left(t_b \right) \frac{\partial \varphi(x(t_b))}{\partial x_2} \Rightarrow \sin \phi_0 = -\cos \theta_d$$

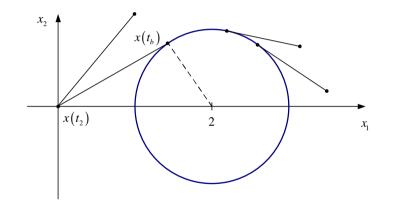
$$\Rightarrow \phi_0 = \pi/6, \quad \theta_d = 2\pi/3$$



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OBSTACLE AVOIDANCE

EXAMPLE: OPTIMAL TRAJECTORIES





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EXAMPLE – MOON LANDER

Consider the decent of a moon lander.

$$\begin{split} \dot{h} &= v \\ \dot{v} &= -g + k \frac{u}{m} \\ \dot{m} &= -u \end{split}$$

The thrust *u* is used to steer the system to h = 0, v = 0. In addition we wish to minimize the fuel used during landing, i.e.

$$J = \int_0^t |u| \, dt$$

or, minimize time,

$$J = \int_0^t dt$$

Control constraint: $0 \le u \le c$, state constraints: $h \ge 0$, m > 0



MOON LANDER

MOON LANDER, NECESSARY CONDITIONS, MIN TIME

$$H(x, u, \lambda) = 1 + \lambda_1 x_2 + \lambda_2 \left(-g + \frac{k}{x_3}u\right) + \lambda_3 u$$

From which,

$$u^* = \arg\min_{u} \left[\left(\lambda_2 + \lambda_3 \frac{x_3}{k} \right) u \right] = cu_0 \left(- \left(\lambda_2 \frac{k}{x_3} + \lambda_3 \right) \right) \\ H^* = \lambda_1 x_2 - \lambda_2 g + \left(\lambda_2 + \lambda_3 \frac{x_3}{k} \right) cu_0 \left(- \left(\lambda_2 + \lambda_3 \frac{x_3}{k} \right) \right)$$

and

$$\begin{aligned} \dot{\lambda}_1 &= 0\\ \dot{\lambda}_2 &= \lambda_1\\ \dot{\lambda}_3 &= -\lambda_2 \frac{k}{x_3^2} c u_0 \left(- \left(\lambda_2 + \lambda_3 \frac{x_3}{k} \right) \right) \end{aligned}$$



MOON LANDER, NECESSARY CONDITIONS, MIN FUEL

$$H(x, u, \lambda) = |u| + \lambda_1 x_2 + \lambda_2 \left(-g + \frac{k}{m} x_3\right) + \lambda_3 u$$

From which,

$$u^* = \arg\min_{u} \left[|u| + \left(\lambda_2 \frac{k}{x_3} + \lambda_3\right) u \right] = cu_0 \left(- \left(1 + \lambda_2 \frac{k}{x_3} + \lambda_3\right) \right)$$
$$H^* = \lambda_1 x_2 - \lambda_2 g + \left(1 + \lambda_2 \frac{k}{x_3} + \lambda_3\right) cu_0 \left(- \left(1 + \lambda_2 \frac{k}{x_3} + \lambda_3\right) \right)$$

$$\begin{aligned} \dot{\lambda}_1 &= 0\\ \dot{\lambda}_2 &= \lambda_1\\ \dot{\lambda}_3 &= -\lambda_2 \frac{k}{x_3^2} c u_0 \left(-\left(1 + \lambda_2 \frac{k}{x_3} + \lambda_3\right) \right) \end{aligned}$$

